# SIMULATION ASSESSMENT OF OPTIMISATION POSSIBILITIES OF COOPERATION OF THE TRACTION ENGINE AND THE TURBOCHARGER

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Summary. The paper presents a selection method of the turbocharger to the engine, which allows for obtaining the maximum efficiency of the supercharging system for the selected engine operation point on the external characteristics. The values of the characteristics of the compressor and the turb ine have been determined on the basis of simulation studies using own numerical program. The calculations have been carried out for the SW 680 engine with the factory B4A turbocharger

Key words: combustion engine, modelling, turbocharging

### INTRODUCTION

One of the main requirements when matching a turbocharger to the engine is to ensure proper efficiency and stability in cooperation with the engine, while significantly reducing the temperature of the medium introduced into the cylinder and the work of the charge exchange and increasing the overall efficiency of the engine. This is also facilitated by proper shaping of the charge exchange system [Lisowski 2008 a, Lisowski 2008 b, Mysłowski 2002]. Optimisation of co-operation of the engine and the supercharging device is important both in terms of the fuel consumption and the dynamic and ecological properties of engines, and is particularly important in traction diesel engines operating in the conditions of high excess air. This is shown by the development trends of new designs of charge compression systems [Christmann et al. 2005, Mährle et al. 2007, Steinparzer et al. 2005, Syomin et al. 2010]. The crucial factor is the wide spread use of the research me thods based on reliable mathematical models of flow processes in the engine and the turbocharger [Sorenson et al. 2005, Winkler et al. 2007, Westin et al. 2005, Vavra et al. 2009].

The quality of the selection of the compressor is estimated on the basis of the location of points of the engine operation, plotted onto the flow characteristics of the compressor. Such characteristics present a dependence of compressor ratio  $\pi_r = f\left(n_{\max}, \dot{m}_{\max}\right)$  and the compression efficiency  $\eta_r = f\left(n_{\max}, \dot{m}_{\max}\right)$  as a function of air mass flow rate  $\dot{m}_{\max}$  and the turbocharger speed  $n_{\max}$  corrected to standard conditions [Wisłocki 1991, Moraal et al. 1999]. With the required values of the supercharging pressure and the air flow determined by the demand for air by the engine, the points of the engine operation should be in the characteristics area of the compressor with a high compression efficiency and within an appropriate distance from the pumping limit.

The mathematical description of the turbine [Macek et al. 2003, Moraal et al. 1999, Wislocki 1991, Wanszejdt et al. 1977] is most often based on the experimentally obtained dimensionless characteristics that are presented in the form of the functional dependence of the scaled mass flow parameter  $F_p = \dot{m}_t \cdot \sqrt{T_t^*} / p_t^* = f \left( n_o / \sqrt{T_t^*}, \pi_t^* \right)$  on the turbine speed parameter  $n_o / \sqrt{T_t^*}$  ( $u_t / \sqrt{T_t^*}$ ) and the expansion ratio  $\pi_t$  and the adiabatic efficiency dependence  $\eta_{at} = f \left( n_o / \sqrt{T_t^*}, u_t / c_{ad} \right)$  on the blade-speed parameter  $u_i / c_{ad}$  and the parameter of  $n_o / \sqrt{T_t^*} = (u_t / \sqrt{T_t^*})$ . The assessment of the turbine matching to the traction engine based on the available flow characteristics is, however, much more difficult. Such characteristics obtained on flow stands in the conditions of a steady flow of the stream of gas allow for an assessment of the degree of perfection of a particular design. They do not allow, however, for reliable assessment of the quality of their cooperation with the engine. In the case of traction engines, where there are used systems with pulsation supply one must take into account the impact of the pulsatility of the exhaust stream to the change in the efficiency and throughput of the turbine [Macek et al. 2005, Westin et al. 2005, Vavra et al. 2009].

### PROBLEM DESCRIPTION

Ensuring of the best working conditions of the turbine operation requires keeping the optimum value of the ratio of the peripheral speed u of the turbine rotor to the gas flow velocity through the turbine  $c_{sc}$  at the adiabatic expansion  $u_{t}/c_{sc}$   $\left(u_{t}/c_{sc}\right)_{agc}$  that should correspond to the maximum turbine efficiency  $\eta$ . In the systems with pulsation turbine supply, for assessment of the actual conditions of the turbine operation it is necessary to determine the flow velocity of the exhaust gases stream  $c_{sc}$  with taking into account the energy of the pressure pulses supplying the turbine. In such conditions, the average velocity  $c_{sc}$  significantly differs from the values obtained in a steady flow. The criterion for optimum matching of the characteristics of the turbine with pulsed supply to the engine at the velocity of  $c_{sc}$  may be presented in the form of the dependence:

$$u_t/c_{ad} = (u_t/c_{ad})_{ad}$$
 (1)

With a known stream  $m_t$  and the curve of the variation of the pressure and the temperature of exhaust gases, the velocity  $c_{nlm}$  (for any point of the engine operation determined by the speed parameter value  $n_n/\sqrt{\Gamma_t^*}$ ) can be determined on the basis of the averaged, within one working cycle of the engine (720 degrees of crankshaft rotation) value of the stream of the exhaust gas enthalpy [Grodzijewskij 1986, Korczewski 2004]:

$$c_{\text{ad-av}} = \frac{2 \cdot R_{\text{av}} \cdot \int_{a_{t}}^{a_{t}} \tilde{m}_{t} \cdot T_{t}^{*} \cdot \frac{\kappa_{t}}{\kappa_{t} - 1} \cdot \left[ 1 \cdot \left( \frac{1}{\pi_{t}^{*}} \right)^{\frac{\kappa_{t} - 1}{\kappa_{t}}} \right] d\alpha}{\int_{a_{t}}^{a_{t}} \tilde{m}_{t} d\alpha}.$$
(2)

The impact of the pulsatility of the exhaust gases stream on the parameters of the turbine operation can be presented by the so-called pulsatility coefficients. The values of these coefficients can be determined based on the results of identification research of a particular engine [Kowalczyk et al. 1989] or can be taken from the literature [Wanszejdt et al. 1977]. The coefficients presented there have been arranged according to the criterion of similarity of the exhaust gases flow in the

exhaust pipes of engines with similar design to with specified number of cylinders connected to one outlet pipe.

The coordinates of each point of the engine operation in the characteristics of the compressor correspond to the specified rotational speed of the turbocharger and the value of the turbine speed parameter  $n_{\omega}/\sqrt{T_{\epsilon}^{*}}$  ( $u_{\epsilon}/\sqrt{T_{\epsilon}^{*}}$ ), depending on which the values of the characteristics of the turbine are determined. Thus, the value of the parameter  $n_{\omega}/\sqrt{T_{\epsilon}^{*}}$  ( $u_{\epsilon}/\sqrt{T_{\epsilon}^{*}}$ ), that determines the required conditions of the turbine operation is to be taken as set, depending on the point of the compressor selection. In this case, for the calculated value of the flow parameter  $F_{\mu}$  the criterion of the optimum matching of the turbine is reduced to compliance with the condition:

$$u_{i}/c_{ad-av} = \left(u_{i}/c_{ad}\right)_{cor}$$
 (3)

The appropriate value of the speed  $u_i$  can be achieved by changing of the turbine blade diameter  $D_{rp}$  while  $c_{nd,rp}$ —as a result of the selection of effective turbine flow area  $A_{rp}$ . In the case of a single-stage supercharging without regulation, a significant restriction for the optimisation of the parameter  $u_i/c_{nd}-c_{np}$  is the condition of not exceeding the permissible value of the supercharging pressure at the rated engine operation point. This condition determines the need to keep a determined value of the effective turbine flow area  $A_{rp}$ . Depending on the selection of the turbocharger, compliance with the condition (3) can be obtained at the speed corresponding to the maximum torque, or for the rated engine operation conditions, or at any other point of its characteristics.

### SIMULATION RESEARCH

For assessment of the conditions for cooperation of the engine and the turbocharger an own computational model has been used, based on mathematical models of particular components of the system: diesel engine, radial compressor, radial turbine [Danilecki 2007]. The identification of the model coefficients has been carried out on the basis of the results of measurements carried out for the SW 680 engine. The design of the model allows for determination of the average values of the circulation of the engine with a selected turbocharger set for the given load value and the speed. In this model, changes in the turbine power and throughput with pulse exhaust gases flow are taken into account by means of the pulsatility coefficients  $k_{\rm w}$  and  $k_{\rm r}$  [Wanszejdt st al. 1977]. The power balance equation for the compressor N, and the turbine N, with taking into account the pulsatility coefficient  $k_{\rm w}$  is defined by the dependence:

$$N_{s} = k_{s} \cdot N_{s}. \tag{4}$$

The values of the pulsatility coefficient  $k_{rr}$  are determined from the empirical functional dependence  $k_{rr} = f(\pi_{rr}, \pi_{rr})$  from the compression of the compressor  $\pi_{rr}$  and the expansion ratio of the turbine  $\pi_{rr}$ .

Changes of the stream of the exhaust gases mass m, with pulsating flow through the turbine, which are taken into account by the coefficient k, are determined by the expression:

$$F_{g} \cdot k_{f} = \frac{m_{f} \cdot \sqrt{T_{c}^{r}}}{p_{f}^{r}}$$
 (5)

The coefficient k, is calculated from the formula:

$$k_f = 1/\sqrt{k_N}$$
. (6)

The average velocity of the stream c is determined from the dependence:

$$c_{ad,av} = \sqrt{2 \cdot H_t^{ad}} \text{ [m/s]}.$$
 (7)

The enthalpy of the exhaust gases at the expansion ratio determined with taking into account the pulsatility coefficient  $k_r(5)$  is calculated from the formula:

$$\mathbf{H}_{t}^{\text{ad}} = \frac{\kappa_{t}}{\kappa_{t} - 1} \mathbf{R}_{\mathbf{q}_{t}} \cdot \mathbf{T}_{t}^{*} \cdot \left[ 1 - \left( \frac{1}{\pi_{t}^{*}} \right)^{\frac{\kappa_{t} - 1}{\kappa_{t}}} \right] [J/kg]. \tag{8}$$

To determine the expansion ratio in the turbine  $\pi_t = p/p_n$  with the calculated stream of exhaust gases (5) the turbine flow characteristics is used  $F_e = \hat{m_e} \cdot \sqrt{\Gamma_e^*} / \hat{p_e^*}$ . The values  $F_p$  are determined depending on the expansion ratio  $\pi_e$  and the turbine speed parameter  $u_e / \sqrt{\Gamma_e^*}$ , while the adiabatic efficiency  $\eta_{ad}$  from the dependence simplified to the form of  $\eta_{ad} = f(u/c_{ad})$ .

Using the developed computational model that allows for determination of the course of the engine and turbocharger characteristics, relevant re-calculations have been carried out for different values of the blade diameter  $D_{\tau}$  and the flow area  $A_{\tau}$  of the turbine. For the value of the flow parameter  $F_{\rho}$  determined from the turbine characteristics, such values  $D_{\tau}$  and  $A_{\tau}$  of the parameters have been sought, at which the conditions of the engine supercharging are kept at the point of the compressor selection, and where the optimisation criterion is met (3).

Comparative calculations have been carried out for the SW 680 engine, which is equipped at the factory with the B4A turbocharger, with the compressor rotor with denotation of 270 and the turbine with the blade diameter of  $D_{\tau}=94$  mm, and a effective turbine flow area  $A_{\tau}$  of 21 cm<sup>2</sup>. The co-operation of the engine and the turbocharger has been analysed in the conditions of the external characteristics and the load characteristics. Calculations in the conditions of the external characteristics have been carried out within the range of the rotational speed of 1200 rpm at  $p_{e}=0.974$  MPa up to the speed of 2200 rpm at  $p_{e}=0.749$  MPa, which correspond to the rated power of the engine. The load characteristics have been determined at the speed of 1400 rpm within the range of variability of the average effective pressure from  $p_{e}=0.55$  MPa up to  $p_{e}=0.985$  MPa, corresponding to the maximum torque value.

### DISCUSSION OF RESULTS

It results from the obtained calculation results (Fig. 1) that in all the considered points of the engine operation, the turbine operates at the values of the parameter  $\mathbf{u}/c_{\mathbf{u}-\mathbf{v}}$  lower than the optimum that for the most of small-scale turbochargers designed for traction engines is virtually constant, and is approximately 0.7 [Moraal *et al.* 1999, Mysłowski 2002]. In such conditions, also the efficiency of the turbine  $\eta$ , is lower than its maximum value. The most

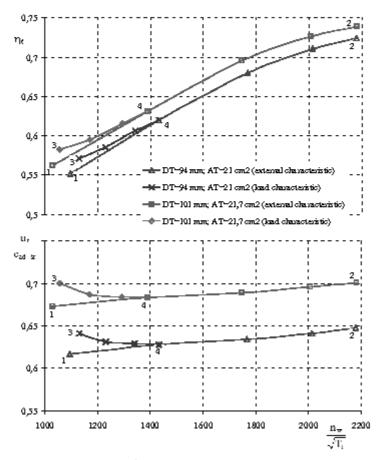


Fig. 1. Changes in the parameter  $u/v_{cal-m}$ , depending on the blade diameter  $D_r$  and the flow area  $A_r$  of the turbine of the SW 680 engine in the conditions of the external characteristics and the load characteristics; (1-at n = 1200 rpm at  $p_r = 0.974$  MPa, 2-n = 2200 rpm at  $p_r = 0.749$  MPa, 3-n = 1400 rpm at  $p_r = 0.985$  MPa)

favourable conditions of the turbine operation are close to the rated operation point of the engine and in the conditions of the load characteristics (points 2 and 3, Figure 1), where  $u_r/c_{nd-n}$  reaches the highest values. To ensure optimum operation of the turbine both at the rated power and at the partial loads of the engine, one should increase the value of the parameter  $u_r/c_{nd-n}$ , from the value of 0.647 up to 0.7. Simple re-calculations show that this can be achieved by increasing the diameter  $D_r$  of the turbine rotor from 94 to approximately 102 mm. Such change leads to the desired changes in the peripheral speed of the rotor  $u_r$ , while allowing for keeping a constant rotational speed of the rotor and the required value of the parameter  $n_{rr}/\sqrt{T_r^*}$ . Please note that following the increase in the diameter  $D_r$  also the maximum value of the efficiency of the turbine increases, which results in a certain change in the speed value  $c_{nd-n}$ . As a result, this involves a certain correction of the effective turbine flow area  $A_r$ . The calculated optimum values of these parameters are:  $D_r = 101 \text{ mm}$ ,  $A_r = 21.7 \text{ cm}^2$ . At the same time, the efficiency of the turbine has increased from 0.724 up to 0.739, while maintaining the supercharging conditions at the rated engine operation point (point 2, Figure

2). As a result of the increase in the diameter  $D_{\tau}$  and the flow area  $A_{\tau}$  one can observe shifting of the engine operation points toward the lower values of the compression and the air flow within the range of low and medium speeds and under partial loads (points 1, 3, 4, Figure 2), leading to slight deterioration of supercharging at the maximum torque.

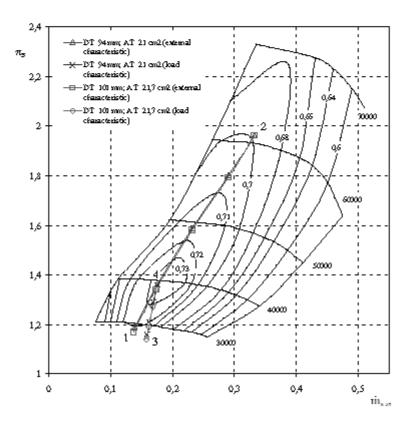


Fig. 2. Changes in the position of the points of the SW 680 engine in the characteristics of the compressor B4A 270 depending on the blade diameter  $D_{\tau}$  and the flow area A, of the engine turb ine in the conditions of the external and load characteristics, (1-n=1200 rpm, 2-n=2200 rpm, 3-n=1400 rpm at  $p_{\epsilon}=0.55 \text{ MPa}$ , 4-n=1400 rpm at  $p_{\epsilon}=0.985 \text{ MPa}$ )

The performed calculations have shown that the optimisation of the conditions of the turbine operation at the maximum torque of the engine would require an increase in the rotor diameter from 94 up to 103 mm, while maintaining the flow area  $A_r = 21 \text{ cm}^2$ . With so selected turbine parameters in the rated conditions, shifting of the engine operation point would take place in the compressor characteristics (point 2, Figure 2) towards higher values of compression, exceeding the imposed limits (increase in the value of the parameter  $n_w / \sqrt{\Gamma_t^*}$ ).

#### CONCLUSIONS

The presented method allows for assessment of the quality of matching the experimentally determined characteristics of the turbine to the flow characteristics of the traction engine. It allows for selection of the turbine design parameters, at which optimum conditions for its cooperation with the engine in the conditions of a pulsating exhaust gases flow will be ensured. Application of this method may be significant both at the optimisation of the existing engines, where the curves of the variations of the exhaust gases pressure are known in the exhaust system, and at the development of new designs, for which the relevant calculations can be performed based on the pulsatility coefficients available in the literature.

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## SYMULACYJNA OCENA MOŻLIWOŚCI OPTYMALIZACJI WSPÓŁPRACY SILNIKA TRAKCYJNEGO I TURBOSPRĘŻARKI

Streszczenie. W pracy przedstawiono sposób doboru turbosprężarki do silnika, który pozwala na uzyskanie maksymalnej s prawności układu doładowania dla wybranego punktu pracy silnika na charakterystyce zewnętrznej. Wartości charakterystyk sprężarki oraz turbiny wyznaczono na podstawie badań symulacyjnych przy wykorzystaniu własnego programu numerycznego. Obliczenia przeprowadzono dla silnika SW 680 z fabryczną turbosprężarką B4A.

Slowa klurzowe: silnik spalinowy, modelowanie, turbodoładowanie